## NASA TECHNICAL Memorandum



N 73 - 24597 NASA TM X-2798

# CASE FILE COPY

PROCESSABLE HIGH-TEMPERATURE-RESISTANT ADDITION-TYPE POLYIMIDE LAMINATING RESINS

by Tito T. Serafini and Peter Delvigs

Lewis Research Center Cleveland, Ohio 44135

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION . WASHINGTON, D. C. . MAY 1973

1. Report No.	2. Government Access	ion No.	3. Recipient's Catalog	No.	
NASA TM X-2798	<u> </u>		5 8 2		
•	4. Title and Subtitle		5. Report Date May 1973		
PROCESSABLE HIGH-TEMPER			6. Performing Organiz	ration Code	
ADDITION-TYPE POLYIMIDE LAMINATING RESINS			o. renoming organiz	ation code	
7. Author(s)			8. Performing Organiz	ation Report No.	
Tito T. Serafini and Peter Delvigs			E-7324		
			10. Work Unit No.		
9. Performing Organization Name and Address			501-21		
Lewis Research Center	Lewis Research Center		11. Contract or Grant	No.	
National Aeronautics and Space Administration					
Cleveland, Ohio 44135		-	13. Type of Report and Period Covered		
12. Sponsoring Agency Name and Address			Technical Memorandum		
National Aeronautics and Space	Administration	F	14 Conseries Assess	Code	
Washington, D. C. 20546			14. Sponsoring Agency	Code	
15. Supplementary Notes					
16. Abstract					
1					
This report reviews basic stud	-	Ŧ.,	-		
polymerization mechanism of t	he so-called addi	tion-type (A-type) p	olyimides. The	fabrication	
and properties of polyimide/gr	aphite fiber comp	osites using A-type	polyimide prep	olymers as	
the matrix are also reviewed.	An alternate met	thod for preparing p	rocessable A-ty	pe polyi-	
mides by means of in situ poly					
ment is described. The elevated temperature properties of A-type PMR/graphite fiber com-					
posites are also presented.					
				•	
].			<i>:</i>		
·					
17. Key Words (Suggested by Author(s))	17. Key Words (Suggested by Author(s))		18. Distribution Statement		
Monomer reactants					
Thermally stable polymers		Unclassified - v		-	
Thermally stable polymers	·			-	
Thermally stable polymers Polyimides	·			-	
Polyimides Composites	lest and				
Polyimides Composites	20. Security Classif. (c	Unclassified - v	ınlimited	- 22. Price*	
Polyimides Composites	20. Security Classif. (d	Unclassified - v		22. Price* \$3.00	

<sup>\*</sup> For sale by the National Technical Information Service, Springfield, Virginia 22151

# PROCESSABLE HIGH-TEMPERATURE-RESISTANT ADDITION-TYPE POLYIMIDE LAMINATING RESINS

by Tito T. Serafini and Peter Delvigs

Lewis Research Center

#### **SUMMARY**

An important finding that resulted from research that was conducted to develop improved ablative resins was the discovery of a novel approach to synthesize processable high-temperature-resistant polymers. Low-molecular-weight polyimide prepolymers end-capped with norbornene groups were polymerized into thermo-oxidatively stable modified polyimides without the evolution of void producing volatile materials.

This report reviews basic studies that were performed using model compounds to elucidate the polymerization mechanism of the so-called addition-type (A-type) polyimides. The fabrication and properties of polyimide/graphite fiber composites using A-type polyimide prepolymers as the matrix are also reviewed. An alternate method for preparing processable A-type polyimides by means of in situ polymerization of monomer reactants (PMR) on the fiber reinforcement is described. The elevated temperature properties of A-type PMR/graphite fiber composites are also presented.

#### INTRODUCTION

In order to synthesize polymers capable of withstanding temperatures of about  $315^{\circ}$  C  $(600^{\circ}$  F) for extended periods of time, it is necessary to incorporate thermo-oxidatively stable aromatic and/or heterocyclic structural units in the polymer molecular structure. Until recently, all thermally stable polymers were synthesized by condensation reactions. These polymers can be referred to as C-type polymers. There are a number of disadvantages associated with the use of C-type polymers as matrix resins for fiber reinforced composites. The primary disadvantage is conversion of the polymer into an intractable state prior to the complete elimination of void-producing volatile materials.

Under NASA sponsorship, TRW, Inc., developed an approach to prepare processable

thermally stable polyimides, known as A-type polyimides, which cure by an addition reaction (ref. 1). This approach utilizes low molecular weight amide-acid prepolymers end capped with reactive norbornene groups that polymerize into thermo-oxidatively stable polyimides without the evolution of byproducts. A method was developed in our laboratory to fabricate A-type polyimide/fiber composites using a technique of in situ polymerization of monomers (ref. 2).

This report reviews composites fabrication and characterization studies that were conducted using the prepolymer or in situ polymerized A-type polyimides.

#### DISCUSSION

Studies performed under NASA sponsorship to develop improved ablative resins were reported by Burns, Lubowitz, and Jones (ref. 1). These investigators concluded that the polymers best able to withstand the combustion environment of fluorine containing propellants were polymers whose molecular structure contained a multiplicity of aromatic groups. Their studies with model compounds such as those shown in figure 1 to determine the temperatures required for thermally induced aromatization led to the discovery of what are known today as addition-type (A-type) polyimides. It can be seen in figure 1 that the group II bis-imide capped with the partially unsaturated bridged structure gave a higher weight retention than either group I (completely aromatic) or

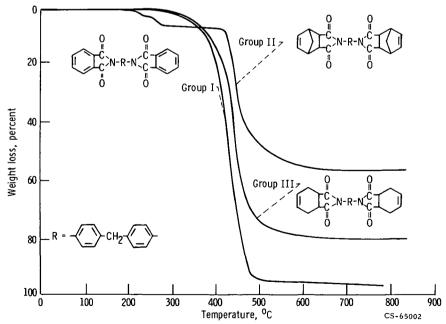


Figure 1. - Thermograms of model bisimides (ref. 1). Environment, nitrogen; scanning rate,  $3^{\rm O}$  C per minute.

group III (partially unsaturated) model compounds. The inflections in the thermogravimetric analysis (TGA) curve between 200° and 300° C for the group II model compounds, compared with the smooth curves for model compounds in groups I and III, led them to conclude that macromolecules were formed during the heating of the group II model compounds. They referred to the polymerization process as pyrolytic polymerization. They immediately realized the significance of this finding and their continued research culminated in the development of the A-type polyimide known as P13N. This processable polyimide met with considerable acceptance by the plastics industry. The structures of the amide-acid and imidized prepolymers of P13N are shown in figure 2. P13N is available from the Ciba-Geigy Corporation which acquired license to the material from TRW, Inc.

Under subsequent NASA sponsored programs (refs. 3 and 4) studies were performed to develop A-type polyimides having improved thermo-oxidative stability (TOS) at 315° C (600° F) for use as laminating resins. Some studies were also performed during these programs to elucidate the mechanisms of the reactions leading to the formation of macromolecules. Pyrolysis studies were conducted with the model compound I, N-phenyl-5-norbornene-2, 3-dicarboximide (fig. 3) under vacuum, pressure, or pressure and catalyst environments. It was concluded that essentially the same polymer structure results from pyrolysis in any of the environments (ref. 4). It should be pointed out that low-molecular-weight oligomers, instead of high molecular weight polymers, resulted from the pyrolysis of the model compound. The postulated reaction mechanism is shown in figure 3. The first step involves a reverse Diels-Alder reaction, which leads to the formation of cyclopentadiene and N-phenylmaleimide, which immediately co-react to form adduct II. It was further postulated that the adduct initiates homopolymerization of the norbornene species I. The various environments only seem to affect the degree of polymerization. Thermogravimetric analyses of residues produced at pyrolysis temperatures in the range of  $271^{\circ}$  to  $350^{\circ}$  C indicated the optimum pyrolytic polymerization temperature to be 316° C.

Figure 4 shows the results from isothermal-gravimetric analysis (ITGA) in air at  $315^{\circ}$  C ( $600^{\circ}$  F) of various A-type polyimides (ref. 3). The label given on each curve identifies the reactants that were used to synthesize the prepolymers, which were subsequently cured to cross-linked polyimides. The stoichiometry of the reactants was selected to prepare prepolymers having an average formulated molecular weight (FMW) of 1300. It is clear from the results presented in figure 4 that the use of 4,4'-oxydianiline (ODA) with either 3,3',4,4'-benzophenonetetracarboxylic dianhydride (BTDA) or pyromellitic dianhydride (PMDA) resulted in A-type polyimides exhibiting lower TOS than the use of 4,4'-methylenedianiline (MDA). The most significant finding was that the polymers made from NA/MDA/PMDA prepolymer exhibited higher TOS than any of the other A-type polyimides including P13N (curve labeled NA/MDA/BTDA in fig. 4).

Amide-acid prepolymer

Figure 2, - P13N prepolymers.

Figure 3. - Pyrolytic polymerization reaction mechanism (ref. 4).

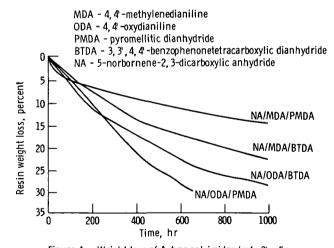


Figure 4. - Weight loss of A-type polyimides (ref. 3). Environment, air at  $315^0$  C  $(600^0$  F); formulated molecular weight, 1300.

Composite fabrication studies with the NA/MDA/PMDA prepolymers showed that it was necessary to adjust the stoichiometry of the reactants to yield prepolymers having an average FMW of 1000. This prepolymer became known as P10P. The reduction in the FMW was necessary for adequate resin flow. Properties of composites made from the P10P prepolymer and Hercules HTS graphite fibers are shown in figure 5 as a function of exposure time in air at 260° and 315° C (500° and 600° F) (ref. 5). The mechanical property results presented in figure 5 clearly show excellent retention of properties at 260° C (500° F) for 1000 hours. In contrast, at 315° C (600° F) after 600 hours of exposure the composites retained approximately 30 percent of their room temperature values. It was suggested (ref. 5) that some of the composite property degradation at 315° C (600° F) could be attributed to oxidative degradation of the HTS graphite fiber.

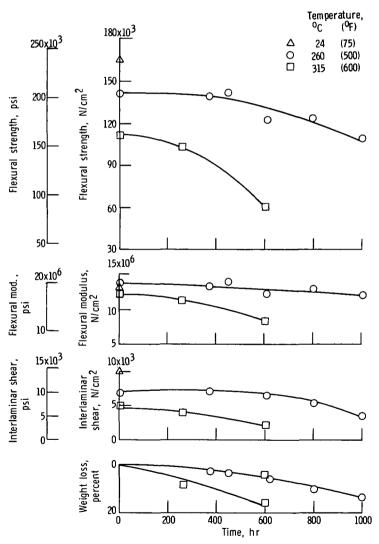


Figure 5. - Properties of HTS/P10P composites as function of time at temperature (ref. 5).

TABLE I. - STRUCTURES OF ESTER ACIDS

Structure	Name .	Abbreviation
C-OMe	Monomethyl ester of 5-norbornene-2, 3-dicarboxylic acid	NE
MeO-C 0 C-OMe HO-C 0 C-OH	Dimethyl ester of 3,3',4,4'- benzophenonetetracarboxylic acid	BTDE
MeO-C C-OMe 0 0 C-OMe 0 0	2,5-dicarbomethoxyterephthalic acid	PMDE

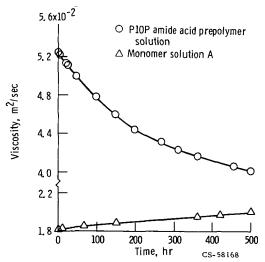


Figure 6. - Viscosity changes of P10P amide acid and monomer solution A (ref. 2). 20 weight percent solids in DMF at 25<sup>0</sup> C.

To prepare A-type polyimides, another approach was developed in our laboratories (ref. 2). Our approach eliminates the need for prepolymer synthesis and circumvents some of the shortcomings of A-type amide-acid prepolymers. The method involves the use of ester-acids rather than anhydrides, thereby preventing the formation of prepolymers at room temperature. A solution containing a diamine and the ester-acids is used to impregnate the reinforcing fibers. In situ polymerization of the monomer reactants (PMR) occurs upon heating the impregnated fibers. Table I shows the structures of the various ester-acids used in the study reported in reference 2. Figure 6 shows the variation of solution viscosity with time for monomer and prepolymer solutions. Monomer solution A (fig. 6) has the same stoichiometry as the amide-acid prepolymer P10P. Figure 6 shows the following: (1) Monomer solutions exhibit considerably less viscosity variation than prepolymer solutions, and (2) monomer solutions have a much lower viscosity. Their lower viscosity enhances fiber wetting and also permits the use of solutions having higher solids contents. In contrast to amide-acid prepolymers (which require the use of high boiling aprotic solvents, such as N, N-dimethylformamide), low boiling alcohols, that is, methanol, can be used as solvents for monomer solutions.

The variation of composite interlaminar shear strength with exposure time in air at 315°C (600°F) for HTS graphite composites made from either a monomer or prepolymer solution is shown in figure 7. It can be seen that the interlaminar shear strength retention of the PMR/HTS fiber composites was essentially identical to the interlaminar shear strength retention of the P10P/HTS fiber composites. Comparison of other properties, such as flexural strength, established the utility of the PMR approach. In fact, a composite made from a NE/MDA/BTDE solution, in which the stoichiometry had been selected to yield a prepolymer having an average FMW of 1500, exhibited a 9.5-percent

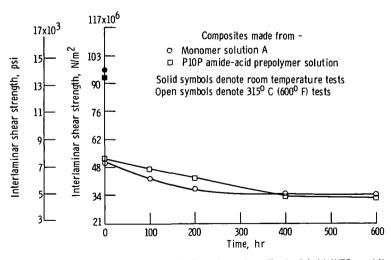


Figure 7. - Interlaminar shear strength of polyimide/ HTS graphitefiber composites (ref. 2).

weight loss after 600 hours in air at 315° C (600° F) (ref. 2). Under identical conditions the P10P/HTS composite weight loss was approximately 18 percent.

Studies conducted to establish the validity of the aforesaid finding and to improve the TOS of PMR A-type polyimides are reported in reference 6. In the earlier work using the PMR approach, the diamines used in the prepreg solutions were either MDA or TDA (ref. 2). Six additional diamines were used in a composite screening study reported in reference 6. Their structures are shown in table II.

TABLE II. - STRUCTURES OF DIAMINES

Structure	Name	Abbreviation
$H_2N - CH_2 - NH_2$	4,4'-methylenedianiline	MDA
$H_2N \longrightarrow 0 \longrightarrow NH_2$	4,4'-oxydianiline	ODA
$H_2N - S - NH_2$	4, 4'-thiodianiline	TDA
H <sub>2</sub> N-\( \sigma \) SO <sub>2</sub> -\( \sigma \) NH <sub>2</sub>	4,4'-sulfonyldianiline	SDA
H <sub>2</sub> N-\_NH <sub>2</sub>	Benzidine	BZD
H <sub>2</sub> N -CH=CH-CH-NH <sub>2</sub>	4, 4'diaminostilbene	DAS
H <sub>2</sub> N-\(\sum_\) NH <sub>2</sub>	<u>p</u> -phenylenediamine	PPDA
H <sub>2</sub> N-\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\	<u>m</u> -phenylenediamine	MPDA

The chemical composition and some properties of 11 different composites are summarized in table III. The stoichiometry of the reactants was governed by the desired number of imide rings or by the desired percent of alicyclic content. The formulated molecular weight is considered to be the average molecular weight of the imidized prepolymer that could have been formed if amide-acid prepolymers had been synthesized. For any given monomer combination, the molar ratio of 5-norbornene-2, 3-dicarboxylic acid monomethyl ester/diamine/tetracarboxylic acid dimethyl ester was 2/(n+1)/n. The composite processing conditions used throughout this investigation were essentially those that had been optimized for amide-acid prepolymer solutions (ref. 3).

Samples 1 to 5 are chemically similar to the commercially available P13N resin. All of them, however, have a lower alicyclic content than P13N. Samples 1 to 3 possessed a combination of unusually high thermo-oxidative stability and high initial interlaminar shear at room temperature. The results from a more detailed study of their mechanical properties after exposure at  $315^{\circ}$  C ( $600^{\circ}$  F) will be described in a subsequent section of this paper.

Sample 6 has a composition equivalent to that of P10P prepolymer. Its properties

Sample Monomer Average Formulated Laminate properties Composite number solution number molecular weight Interlaminar shear Interlaminar shear composition of repeat weight, loss, at 24° C (75° F) at 315° C (600° F) percent<sup>a</sup> **FMW** units, n  $N/m^2$  $N/m^2$ ksi ksi 105.5×10<sup>6</sup> 41.4×10<sup>6</sup> 1 NE/MDA/BTDE 2.087 1500 15.3 6.0 9.5 2 2.603 1750 99.3 14.4 37.9 5.5 9.8 15.0 5.1 3 2.913 1900 103.4 35.2 10.0 3.120 2000 67.6 9.8 31.7 4.6 4 11.4 5.186 3000 56.5 5 8.2 35.2 5.1 28.0 6 NE/MDA/PMDE 10.8 6.2 1.342 1000 74.5 42.7 17.6 NE/MDA/PMDE 1.868 1200 84.8 12.3 42.7 6.2 17.9 7 NE/MDA/PMDE 8 2.658 1500 42.1 6.1 20.0 2.9 20.8 NE/MDA/ 2.338 1500 98.6 14.3 42.7 6.2 10.6 (1BTDE:1PMDE) 10 NE/MDA/ 2.917 1750 93.8 13.6 37.9 5.5 8.6 (1BTDE:1PMDE) NE/MDA/ 1900 12.6 31.0 4.5 10.9 11 3.264 86.9 (1BTDE:1PMDE)

TABLE III. - PROPERTIES OF POLYIMIDE/HTS GRAPHITE FIBER COMPOSITES

<sup>&</sup>lt;sup>a</sup>After 600 hr in air at  $315^{\circ}$  C (600° F).

were shown to be virtually identical with those of composites made from P10P. Samples 7 and 8 have the same chemical composition as sample 6, but differ in stoichiometry.

The monomer combination used in samples 9 to 11 was selected to investigate the effect of increasing the number of imide linkages over that in the corresponding samples 1 to 3 while maintaining a constant alicyclic content. For example, sample 9 has the same alicyclic content as sample 1, but a higher number of imide linkages. The higher number of imide linkages is achieved by partial substitution of PMDE for BTDE. Theoretically this should give a correspondingly higher thermo-oxidative stability. Actually, the results indicate that the thermo-oxidative stabilities of samples 9 to 11 are approximately equivalent to those of samples 1 to 3.

Results from the screening study with diamines other than MDA showed that the incorporation of these diamines did not improve composite properties. In some instances, for example, with PPDA, BZD, and DAS, the composites were poorly consolidated and could not be tested.

Based on the results of the screening study and detailed investigation of samples 1 to 3, 7, and 9, the monomer system that displayed the best overall balance of processability and thermomechanical properties was NE/MDA/BTDE in which n has a value of 2.087 (sample 1, FMW = 1500). Some results of the studies conducted with monomer systems having FMW's of 1500, 1750, 1900, and 1000 (samples 1 to 3, and 6, table I) are presented in figure 8. The range of interlaminar shear strength values is 95 to 123 newtons per square meter (13 800 to 17 800 psi). These values are somewhat higher than those previously reported (ref. 5) for A-type polyimide/HTS fiber composites. There is a considerable drop in the interlaminar shear strength when the composites are tested at  $315^{\circ}$  C ( $600^{\circ}$  F), to as low as 34 newtons per square meter (5000 psi). There was a greater decrease for samples 1 to 3 than for sample 6. The interlaminar shear strength decrease for sample 6 is nearly identical to that reported by Hanson and Serafini (ref. 6).

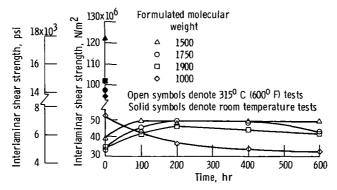


Figure 8. - Interlaminar shear strength of polyimide/ HTS graphite fiber composites (ref. 6).

It appears that composite samples 1 to 3 underwent a postcure on exposure in air at  $315^{\circ}$  C ( $600^{\circ}$  F), since the interlaminar shear strength increased during exposure, whereas that of sample 6 decreased with time. The important fact to note is that the interlaminar shear strength of samples 1 to 3 remained nearly constant from 100 to 600 hours. The strength after 600 hours for samples 1 to 3 averaged approximately 45 newtons per square meter (6500 psi). The data for samples 1 to 3 represent the best interlaminar shear retention for A-type polyimide/HTS fiber composites reported to date. The interlaminar shear strength of sample 1 increased to 48 newtons per square meter (7000 psi) after 200 hours of exposure and remained constant up to 600 hours of exposure.

Photomicrographs of a cross section of composite sample 1, before and after exposure, are shown in figure 9. There is no evidence of voids in the composite as a result of fabrication. After exposure at 315°C (600°F) for 600 hours, the composite shows an extremely small number of voids. This indicates that the original void-free composite was very resistant to the oxidative degradation that might occur by a diffusion mechanism. Any degradation that did occur, took place on the surface of the composite.

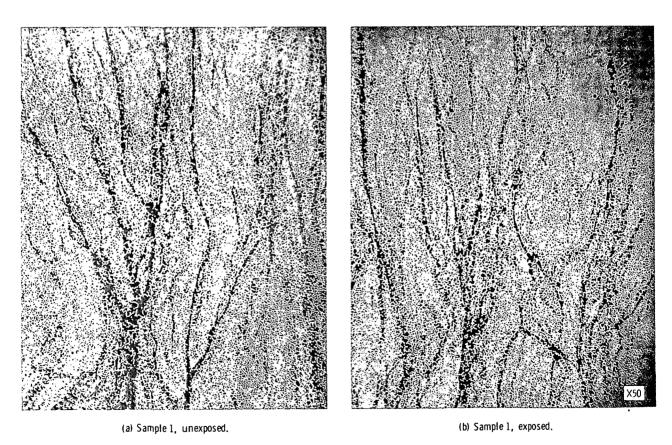


Figure 9. - Polyimide/HTS graphite fiber composites before and after exposure in air at 315° C (600° F) for 600 hours (ref. 6).

The absence of internal voids after exposure also indicates that the composite is very resistant to thermal decomposition.

Under NASA contract NAS3-15829, TRW, Inc., is using the PMR approach to develop high-temperature-resistant polyimide/graphite fiber air breathing engine components, such as blades and vanes. The monomer system being investigated is NE/MDA/BTDE at FMW's in the range of 1500 to 1900. Figure 10 shows the average weight loss of HMS graphite fiber composites exposed at 315°C (600°F) in air. Also shown in the figure is the weight loss of composites fabricated with HTS fiber (ref. 6). It can be seen that the HMS composites exhibited excellent weight retention (~96 percent). The greater weight loss of the HTS composite can be attributed to the lower thermo-oxidative stability of the HTS fiber (ref. 5). The point that can be made is that the investigators were able to fabricate thermo-oxidatively stable composites using the in situ PMR approach with little or no difficulty.

Figure 11 is a flow chart of the PMR approach for the fabrication of highperformance-resin/fiber composites. We have been able to use the PMR approach for
the fabrication of polyphenylquinoxaline/graphite fiber composites. These studies will
be described in a forthcoming publication. The PMR approach as outlined in figure 11
is considerably less complex than composites fabrication processes that use conventional prepreg fabrication techniques. In addition to the advantages discussed previously,
such as low viscosity, high solids content solutions, etc., the PMR approach eliminates
the problems associated with packaging, shipping, and storage of prepreg. The PMR
approach also should ultimately lead to overall cost savings.

We view the PMR approach as one that should permit the use of a number of polymers that have demonstrated outstanding high-temperature TOS but could not be used as matrix resins because of the intractability inherent in their polymer structure.

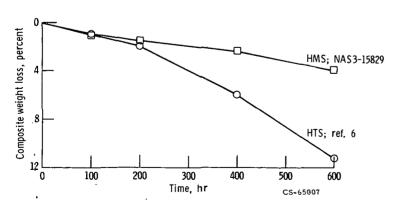


Figure 10. - Weight loss of A-type polyimide/graphite fiber composites (NE/MDA/BTDE). Formulated molecular weight, 1500 to 1900; environment, air at 315° C (600° F).

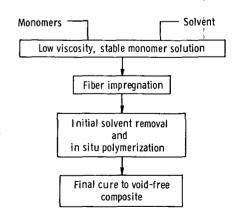


Figure 11. - Approach for preparation of resin/fiber composites using in situ polymerization of monomer reactants.

#### **CONCLUSIONS**

Modification of the molecular structure of aromatic polyimide precursors by incorporating norbornene end groups is an excellent approach to achieve processable thermally stable polyimides for use as matrix resins in fiber reinforced composites. Preparation of A-type polyimide/fiber composites using a technique of in situ polymerization of monomer reactants (PMR) is a powerful approach that should be applicable to a variety of high-temperature polymers. The PMR approach offers a number of advantages to the composites fabricator, such as the use of low boiling, relatively nontoxic solvents, and simplified materials handling logistics.

#### Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, March 26, 1973, 501-21.

#### REFERENCES

- 1. Burns, E. A.; Lubowitz, H. R.; and Jones, J. F.: Investigation of Resin Systems for Improved Ablative Materials. Rep. TRW-05937-6019-RO-OO, TRW Systems Group (NASA CR-72460), Oct. 1, 1968.
- 2. Serafini, T. T.; Delvigs, P.: and Lightsey, G. R.: Thermally Stable Polyimides from Solutions of Monomeric Reactants. J. Appl. Polym. Sci., vol. 16, no. 4, Apr. 1972, pp. 905-916.
- 3. Burns, E. A.; Jones, R. J.; Vaughan, R. W.; and Kendrick, W. P.: Thermally Stable Laminating Resins. Rep. TRW-11926-6013-RO-OO, TRW Systems Group (NASA CR-72633), Jan. 17, 1970.
- 4. Jones, R. J.; Vaughan, R. W.; and Burns, E. A.: Thermally Stable Laminating Resins. Rep. TRW-16402-6012-RO-OO, TRW Systems Group (NASA CR-72984), Feb. 7, 1972.
- 5. Hanson, M. P.; and Serafini, T. T.: Effects of Thermal and Environmental Exposure on the Mechanical Properties of Graphite/Polyimide Composites. NASA TN D-6604, 1971.
- 6. Delvigs, Peter; Serafini, Tito T.; and Lightsey, George R.: Addition-Type Polyimides from Solutions of Monomeric Reactants. NASA TN D-6877, 1972.

OFFICIAL BUSINESS
PENALTY FOR PRIVATE USE \$300

SPECIAL FOURTH-CLASS RATE BOOK POSTAGE AND FEES PAID NATIONAL AERONAUTICS AND SPACE ADMINISTRATION 451



POSTMASTER:

If Undeliverable (Section 158 Postal Manual) Do Not Return

**&** ×

1 10

"The aeronautical and space activities of the United States shall be conducted so as to contribute... to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

—NATIONAL APRONAUTICS AND SPACE ACT OF 1958

### NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

ű,

TECHNICAL MEMORANIDIMS: Information receiving limited distribution because of preliminary data, security classification, or other reasons. Also includes conference

proceedings with either limited or unlimited distribution.

CONTRACTOR REPORTS: Scientific and rechnical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to ment NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include final reports of major projects, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION

PUBLICATIONS: Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION OFFICE

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Washington, D.C. 20546